# Evaluation of Mechanical Properties and Machinability Analysis of A/ 6061 Hybrid Composite

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Received 02 Jan 2018, Accepted 01 March 2018, Available online 02 March 2018, Vol.6 (March/April 2018 issue)

# Abstract

Aluminium Matrix Composites are extensively used due to their desirable properties like low weight, low cost, high strength to weight ratio, good corrosion resistance, good thermal conductivity and high stiffness. Their applications are diversified in production, thermal, marine and automobile industries. Aluminium is extensively used in ships, aircrafts, cars, electrical wires and household utensils because it is abundant in nature. In the present study, Aluminium alloy Al6061 Hybrid Composites reinforced with Boron carbide and Coconut shell ash are fabricated to replace the individual Aluminium alloy Al6061. For that various tests to determine properties such as strength, hardness, wear and corrosion resistance are conducted on composite samples which make them fit to be used in aircraft window frames by reviewing various literatures. In addition to above, machinability analysis is performed on all the specimens and their surface roughness is measured. Based on the results obtained, we can come to a conclusion that the aluminium composite has superior properties than individual Al6061 alloy.

**Keywords:** Aluminium Matrix Composites; Aircraft; Boron carbide; Coconut shell ash; Machinability; Tensile strength; Hardness; Corrosion resistance

# 1. Introduction

A Composite material can be formed by combining two or more dissimilar materials together. The mechanical properties of the composites will be better than the individual components. The composite materials will be composed of two or more phases. They are matrix phase and reinforcement phase [1].

A Metal matrix composite is one of type of composite in which the matrix phase is predominantly a metal or a metal alloy. The metal is the base material which constitutes the major part and the minor constituents are reinforcements that can be in the form of particles, whiskers, continuous and discontinuous fibres [2]. The reinforcement material will be a ceramic or organic compound. The MMC consists of superior properties such as high strength, high stiffness, high electrical and thermal conductivity, greater resistance to corrosion, oxidation and wear when compared to the base material. A hybrid composite is one which involves more than two constituting materials.

Aluminium Matrix Composites are extensively used due to their desirable properties like low weight, low cost, high strength to weight ratio, good corrosion resistance, good thermal conductivity and high stiffness. Their

\*Corresponding author's ORCID ID: 0000-0002-4858-9032 DOI: https://doi.org/10.14741/ijmcr/v.6.2.1 applications are diversified in production, thermal, marine and automobile industries. To name a few, ships, aircrafts, cars, electrical wires and household utensils use aluminium because it is the most abundant material on earth.

## 2. Selection of materials

# A. Matrix

The matrix material to be used was chosen as Al6061 which is a precipitation hardened aluminium alloy, containing iron, silicon and chromium as its major alloying elements as indicated in Table 1. It has good mechanical properties and exhibits good weldability, good formability and high corrosion resistance [3].

Table	1.Co	mposition	of Al	6061
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Constituents	Percentage
Manganese (Mn)	0.108%
Iron (Fe)	0.125%
Copper (Cu)	0.392%
Magnesium (Mg)	0.970%
Silicon (Si)	0.620%
Chromium (Cr)	0.079%
Others (Total)	0.04%
Aluminium (Al)	97.7%

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# **B.** Reinforcement

The materials selected to be reinforced into the metallic matrix are Boron Carbide having particle size of 150  $\mu$ m and Coconut Shell Ash having particle size of 75  $\mu$ m.

Boron carbide is an industrial ceramic which is the third hardest material next to diamond and cubic boron nitride [4]. It is reinforced in the Al6061 matrix to increase strength, hardness, stiffness, wear resistance and impact strength. Its attractive properties are listed in Table 2.

Table 2 Properties of boron carbide

Properties	Units	Values
Density	g/cm3	2.52
Melting point	°c	2450
Fracture toughness	MPa√m	3.0
Thermal	W/mK	40
Conductivity	VV/IIIK	40
Vickers Hardness		3850
Tensile strength	MPa	350
Elastic modulus	GPa	450

Coconut shell is an agricultural waste and is available in very large quantities throughout the tropical countries of the world. Moreover, coconut is becoming an important agricultural product for tropical countries around the world as a new source of energy-biofuel. Previously, coconut shell was burnt as a means of solid waste disposal which contributed significantly to CO2 and methane emissions. However, as the cost of fuel oil, natural gas and electricity supply has increased and become erratic, coconut shell has come to be regarded as source of fuel rather than refuse.

The coconut shell was first crushed and then burnt for two hours to get charcoal. Then the charcoal was heated in a furnace at  $1300^{\circ}$ C by placing it in a graphite crucible for one hour to get the ash.



Figure 1 Coconut Shell Ash

The obtained silica-rich coconut shell ash having a density of 2.05 g/cm3, is thereafter used as a reinforcement material for the preparation of composites. The chemical composition of the rice husk ash obtained is shown in Table 3. Table 3 Chemical composition of coconut shell ash

Chemical Composition	%
Silicon dioxide (SiO <sub>2</sub> )	55.0
Potassium oxide (K <sub>2</sub> O)	19.2
Ca <sup>2+</sup>	12.9
K <sup>+</sup>	13.1

Both Boron Carbide and Rice Husk Ash are used together as reinforcements, because:

Boron Carbide is expensive

Agricultural wastes along with ceramics give improved mechanical properties

Agricultural wastes can give only marginal increase in strength and hardness, if dispersed in Aluminium matrix individually.

Coconut Shell Ash has low elastic modulus

# 3. Fabrication process

Stir casting is the most popular commercial method of producing aluminum based composites. In this method, pre-heated ceramic particulates are incorporated into the vortex of the molten matrix created by a rotating impeller.

Specime ns	Percentag e of Reinforce ment	Weight % of Al6061	Weight % of B₄C	Weight % of CSA
	0	100	0	0
=	5	95	0	5
=	10	90	0	10
IV	4	96	1	3
V	4	96	2	2
VI	4	96	3	1
VII	8	92	2	6
VIII	8	92	4	4
IX	8	92	6	2

#### Table 4 Compositions of composite specimens

The stir casting process starts with the preheating of graphite crucible in a coal-fired furnace for 20 minutes. The Coconut Shell Ash and Boron Carbide particles were initially preheated separately at a temperature of  $250^{\circ}$ C to remove moisture and to help even distribution within Al6061 alloy. The Al6061 alloy billets were charged into the furnace and heated to a temperature of  $750 \pm 30^{\circ}$ C (i.e) above the liquidus temperature of the alloy to ensure that the alloy melts completely. Now with the help of electrical stirrer, the molten alloy is stirred at a constant speed of 450 rpm to create vortex. The preheated Coconut Shell Ash and Boron Carbide particles are then charged into the melt at constant pour rate and stirring of

the slurry was performed manually for 5-10 minutes. Then the melted matrix and reinforced particles are poured into the preheated mould and the pouring temperature should be maintained at 680°C. After solidification, the composite can be cut into different shapes and sizes as per requirements.

#### 4. Measurement of density

Density is an important physical property of any material. For our composite, measurement of density is even more important as the ultimate aim is to reduce weight of aircraft parts. The density of the composite is related to the individual densities of the matrix as well as reinforced material. (i.e) Al6061 and B4C-Coconut Shell Ash respectively. For measuring density, the composite samples of different compositions are milled and grinded to perfect cubes with specific dimensions as per 'ASTM Standard practice for preparation of test materials A370'.

#### Experimental density

Experimental density or particle density of the composite specimen is calculated using 'Archimedes' principle'. Initially for measuring the density, test samples of different compositions are cut from the specimen, grinded and highly polished before weighing. They are made into perfect cubes having dimension (lxbxt) 10x10x10 mm. The mass of the composite is measured using a highly accurate digital mass balance of unit weight 0.01g.

 $\rho$ mmc = (m /(m-m1)) x $\rho$ H2O

Where,

pmmc is the density of the composite,

'm' is the mass of the composite sample in air,

'm1' is the apparent mass of the same composite suspended in distilled water

'pH2O' is the density of distilled water (at 293K) is 998 kg/m<sup>3</sup>. I = length of the sample in mm.

b = breadth of the sample in mm and t = thickness of the sample in mm.





#### Table 5 Determination of density

Specimen	Mass of the sample in air (g)	Apparent mass of the sample in distilled water (g)	True density (g/cc)
I	20.9	12.95	2.63
11	20.8	11.08	2.14
III	21.5	10.53	1.96
IV	20.3	12.24	2.52
V	21.3	12.71	2.48
VI	21.1	12.98	2.6
VII	20.3	11.77	2.38
VIII	20.3	11.91	2.42
IX	20.6	12.07	2.40

#### 5. Tests conducted

The specimens fabricated by Stir casting is cut and machined into test samples of required shape and dimensions for the conduction of various tests.

# A. Mechanical tests

# (i)Tensile test

Tension means pulling force. The tensile test is done in a Universal Testing (model: M3320; range: 0-40 kN) to determine the tensile strength of the specimen. The specimen to be tested is fastened to the two end-jaws of the UTM. Now the load is applied gradually on the specimen by means of pulling the movable crosshead, till the specimen fractures. The corresponding extension of the composite specimen is also noted.

Ultimate Tensile strength = Maximum load given to specimen/ Area of cross section

#### (ii) Brinell Hardness

Hardness test is done on a specimen to know its ability to resist wear, abrasion and indentation. The Brinell hardness is measured from an indentation produced in the composite by applying a constant load of 500 kgf on steel ball indenter in contact with the surface of the specimen for 10 seconds. The test is conducted as per ASTM standards for Brinell hardness testing E10. The indentation in the specimen surface is calculated from the formula:

Brinell Hardness number = 
$$\frac{2P}{\pi D \left[ D - \sqrt{(D^2 - d^2)} \right]}$$

where,

P = load applied on indenter (500 kgf)D = diameter of steel ball indenter (10 mm)d = diameter of ball impression in mm.

# (iii) Charpy Impact test

Impact strength is the capacity of a material to withstand blows without fracture. The impact test is done as per Standard test methods for notched bar tensile strain Impact test method ASTM E23 in Impact testing machine. In impact test, a notch is cut in the specimen upto 2mm which is struck by a single blow in testing machine. The energy absorbed in breaking the specimen can be measured from the scale provided in the machine. In Charpy test of impact strength, the test sample is fixed horizontally to the machine base and the striking hammer is blown to hit the specimen behind the v-notch.

# (iv) Corrosion test

The corrosion behaviour of the composites is studied by weight loss method using mass loss and corrosion rate measurements in both acidic and basic environments. The corrosion test will be carried out by immersion of the test specimens in 1N HCl (3.6ml in 100ml of distilled water) and 1N NaOH (4g in 100ml of distilled water) solutions which will be prepared following standard procedures. The specimens for the test are cut to size 50×12×12 mm and then mechanically polished with emery papers from 150 down to 600 grit sizes to produce a smooth surface. The samples are de-greased with acetone, rinsed in distilled water, and then dried in air before immersion in still solutions at room temperature (25°C). The solutionto-specimen surface area ratio will be about 150 ml cm<sup>-2</sup>, and the corrosion setups are exposed to atmospheric air for the duration of the immersion test. The weight loss readings will be monitored for a period of 24 hours. The mass loss  $(mg/cm^2)$  for each sample will be evaluated in accordance with ASTM G31 Standard practice for lab immersion corrosion testing of metals following the relation:

m.l = mi – mf

# where,

m.l is the mass loss  $(mg/cm^2)$ , mi is the initial weight (g) and mf is the final weight (g).

Corrosion rate for each sample will be evaluated from the weight loss measurements:

 $C.R = KW/\rho AT$ 

where,

C.R is corrosion rate (mm per year),

W is weight loss after exposure time (mi - mf) (g),

 $\rho$  is the density of the composite (initial mass/volume) (g/cm  $^{3}),$ 

A is the area of the sample (length x breadth of the facing side)  $(cm^2)$ ,

T is time of exposure (hours), and K is a constant equal to 87500.

# 6. Result and tabulation

# Table 6 Tensile Test Results





Figure 3 Tensile test results

Fig 3 shows the effect of weight fraction of various combinations of reinforcement materials on the tensile strength. The results show a fluctuation in tensile strength with the addition of reinforcement particles in which 92%Al+6%B4C+2%CSA shows best result.

## Table 7 Brinell hardness test results



Figure 4 Brinell hardness test results

Fig. 4 shows the effect of weight fraction of various combinations of reinforcement materials on the hardness. The results show a fluctuation in hardness with the addition of reinforcement particles in which 90%Al+10%CSA shows best result.

## Table 8 Charpy impact test results

Specimen	1	Ш	Ш	IV	V	VI	VII	VIII	IX
Impact Strength (J/mm <sup>2</sup> )	0.2	0.475	0.35	0.375	0.425	0.475	0.4	0.425	0.5

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Figure 5 Impact test results

Fig. 5 shows the effect of weight fraction of various combinations of reinforcement materials on the impact strength. The results show a fluctuation in impact energy with the addition of reinforcement particles in which 92%Al+6%B4C+2%CSA shows best result.

Specimen	Ini	tial Weight m <sub>i</sub>	Fir Wei m	ial ight 1 <sub>f</sub>	Mass I m <sub>i</sub> - r	.oss n <sub>f</sub>	Corrosi C <sub>R</sub> =KV	on Rate V/pAT
	Α	В	Α	В	Α	В	Α	В
1	20.9	20.9	20.5	19.7	0.4	1.2	1860.1	4030.3
2	19.2	20.9	18.8	19.7	0.4	1.2	1135.8	3407.3
3	19.7	21.6	19.1	20.3	0.6	1.3	1550.1	3720.2
4	20.4	20.3	19.8	19.1	0.6	1.2	1446.7	2893.5
5	21.4	21.2	20.9	20.1	0.5	1.1	1225.1	2695.2
6	21.4	21.1	20.8	19.9	0.6	1.2	1402.2	2804.5
7	21.5	20.3	21	19.3	0.5	1	1276.5	2553.1
8	20.5	20.3	20	19.2	0.5	1.1	1255.4	2762.0
9	20.8	20.6	20.2	19.6	0.6	1	1012.7	2531.8

# Table 9 Corrosion test results



Figure 6 Corrosion test results

Fig. 6 shows the effect of weight fraction of various combinations of reinforcement materials on the corrosion rate when immersed in both acid(A) and base(B) solutions. Basic solution corrodes more than the acidic solution. The results show a fluctuation in corrosion resistance with the addition of reinforcement particles in which 90%Al+10%B4C shows best result.

# 7. Machinability analysis

## A. Machinability

The term machinability refers to the ease with which a metal can be cut (machined) permitting the removal of the material with a satisfactory finish at low cost.

Materials with good machinability require little power to cut, can be cut quickly, easily obtain a good finish, and do not wear the tooling much; such materials are said to be free machining.

# B. Machinability analysis

Machinability can be difficult to predict because machining has so many variables. There are many factors affecting machinability, but no widely accepted way to quantify it. Machinability characteristics of any tool-work pair is to be judged by:

- magnitude of the cutting forces tool wear or tool life
- surface finish
- magnitude of cutting temperature chip forms

# C. Milling

Milling is the machining process of using rotary cutters to remove material from a workpiece by advancing (or feeding) in a direction at an angle with the axis of the tool. It covers a wide variety of different operations and machines, on scales from small individual parts to large, heavy-duty gang milling operations. It is one of the most commonly used processes in industry and machine shops today for machining parts to precise sizes and shapes.



Figure 7 Vertical milling machine

# D. Material Removal Rate (MRR)

The Material Removal Rate can be defined as the amount of material removed divided by the machining time.

# where

mi – mass of the material before milling (g) mf – mass of the material after milling (g) t – machining time (min)

Specimen	Initial mass, m <sub>i</sub> (g)	Final mass, m <sub>f</sub> (g)	Machining time (min)	MRR (g/min)
I	41.8	40	3.12	0.576
=	40.8	39	3.1	0.58
III	36.4	35	3.12	0.451
IV	39.8	38	3.09	0.582
V	36.5	35	3.11	0.482
VI	37.6	36	3.13	0.511
VII	38.4	37	3.13	0.457
VIII	38.5	37	3.12	0.448
IX	37.4	36	3.11	0.45

### Table 10 Material removal rate

## E. Surface Roughness

Surface roughness cannot be obtained as smooth in any manufacturing process. Some deviations or irregularities will occur. The irregularities on the surface will be in the form of successive hills and valleys which are varying in height and spacing. These irregularities are called as surface roughness or surface finish.

# Surface Roughness Tester

Surface roughness tester is used to measure the surface roughness or surface finish of the materials. The roughness tester measures the roughness value in microns ( $\mu$ ).



Figure 8 Surface roughness tester

Specimen	Trial 1	Trial 2	Trial 3	Surface Roughness (μ)
I	1.418	1.275	1.271	1.321
II	1.71	1.924	1.938	1.857
111	1.301	1.433	1.663	1.465
IV	1.788	1.972	2.054	1.938
V	0.895	0.841	0.873	0.87
VI	0.857	0.892	0.848	0.866
VII	1.246	1.165	1.242	1.218
VIII	2.093	2.301	2.793	2.395
IX	1.62	1.415	1.373	1.469

The composite having composition 95%Al+5%CSA is having the maximum material removal rate and the composite having the composition 90%Al+4%B4C+4%CSA showing the maximum surface roughness.

# Conclusion

After the conduction of various tests and obtaining the results, we can come to a conclusion that the Aluminium Matrix composite which contains both boron carbide and coconut shell ash as its reinforcements has improved mechanical as well as tribological properties when compared to individual aluminium alloy.

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